

# Main differences between V11.4 and V11.4.1

This version is almost the same as the V11.4 one. It corrects these two anomalies:

- It is now possible again to have a maneuver (impulsive or continuous one) at the initial date (or near of it).
- The comment included in the header of ephemeris/events files for the **DATE** parameter is correctly updated, taking into account the right time scale.

# Main differences between V11.3 and V11.4

## Summary

- Compatibility with PATRIUS
- Recurrent date events
- New maneuvers triggering criteriae
- Impulsive maneuvers with orbital parameters criteriae
- Recurrent date events
- New data for Dormand Price 853 integrator
- Additional validity criteria on PRS coefficients
- Altitudes on stop criteria
- Best management of consistency between force model and vehicle panels
- By default attitude law
- Ergonomics change for output variables selection
- New possibilities for plotting
- New output variables
- Possibility to customize its own output variables
- New output modes
- Some new util methods
- Anomaly corrections

## Compatibility with PATRIUS

This version is compatible with the version 4.4 of the [PATRIUS](#) library.

## Recurrent date events

Thanks to the V2.1.1 version of [GENOPUS](#), it is now possible to define date (relative or absolute ones) events with a recurrence (for example each day).

## New maneuvers triggering criteriae

Thanks to the V2.1.1 version of [GENOPUS](#), new criteriae on nodes and apsides (as well as recurrent dates) have been added to the previous ones (relative or absolute dates and AOL)

## Impulsive maneuvers with orbital parameters criteriae

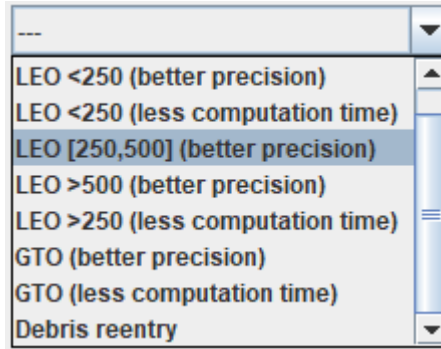
Thanks to the V2.1.1 version of [GENOPUS](#), it is now possible to define an impulsive maneuver giving orbital parameters increments:

- on semi-major axis
- on eccentricity (and semi-major axis eventually)
- on inclination (and semi-major axis eventually)

## New data for Dormand Price 853 integrator

Thanks to the V4.4 version of [PATRIUS](#) and the V2.1.1 version of [GENOPUS](#) with its new specific widget, it is now possible to by-pass the **Dormand Price 853** error mode on the minimum step. When this solution is chosen, the propagation will go on even if the conditions on precision will not be temporary fulfilled.

Moreover, a user help is available to initialize tolerances values for different kind of trajectories:



## Additional validity criteria on PRS coefficients

Thanks to the V2.1.1 version of [GENOPUS](#), besides validity interval of [0,1] for each coefficient, there is an additional test on the sum of the three coefficients (absorption, specular and diffuse) that must be equal to 1.

## Altitudes on stop criteria

It is now possible to choose if the stop criteria will be a geocentric or geodetic altitude.

## Best management of consistency between force model and vehicle panels

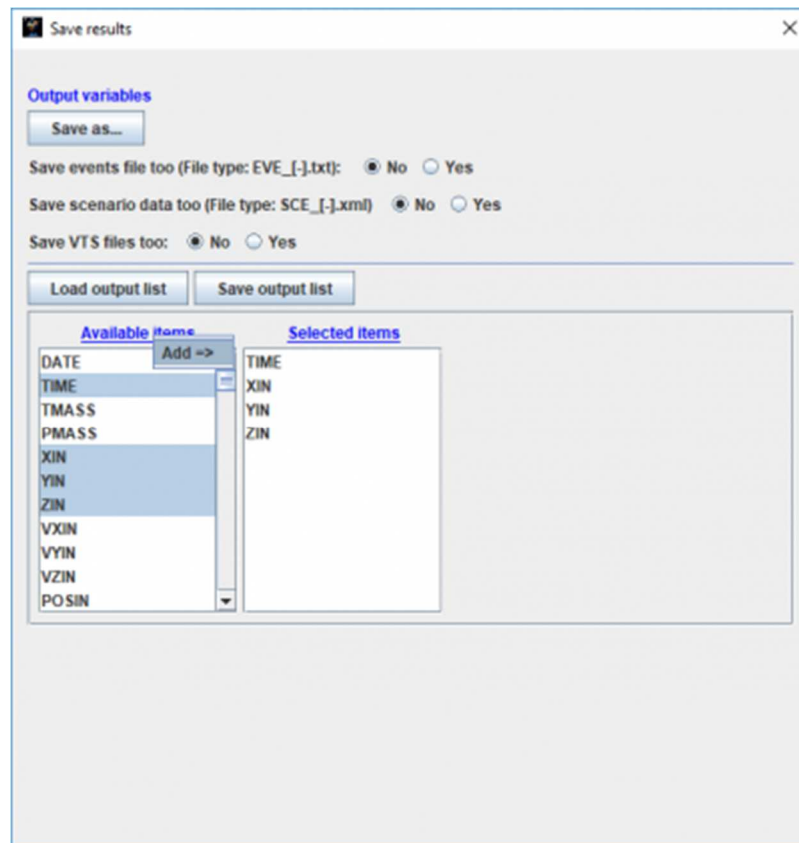
In the previous versions, if we decided to select an atmospheric model without entering data on mass and aerodynamic coefficients, the error was only detected when launching the propagation. Now it is detected sooner and the vehicle tab becomes red as it shifts in error mode.

## By default attitude law

Now an attitude law is defined by default (TNW) to avoid to select if it is not critical.

## Ergonomics change for output variables selection

Thanks to the V2.1.1 version of [GENOPUS](#), we use a [GListSelect](#) widget rather than the previous list.



Using it, it is easier to add / move / remove the variables we want to store in the [MADONA columns format](#) file

## New possibilities for plotting

Thanks to the V1.9.1 version of [GENIUS](#), it is possible to plot dashed curves. Moreover a new utility tool is available outside [PSIMU](#) allowing to plot curves from different files (with possibility of relative information): see [here](#) or getting it with the full [GENIUS package](#).

## New output variables

New variables are now available:

- **F107** (solar Flux / 0 if the atmospheric model does not use it)
- **AP** (geomagnetic index / 0 if the atmospheric model does not use it)
- **DV** (cumulated velocity increment in m/s / 0 if no maneuvers)

Moreover, the time scale used for dates may be selected in the [Output](#) panel (**UTC** by default)

## Possibility to customize its own output variables

**PSIMU** allows to get hundreds of output variables but, for some users, this output list may be incomplete and they will have to wait for a new version to get them. Indeed, these variables could be so particular that there will be no interest to integrate them in a standard **PSIMU** version! A specific mechanism (see [here](#)) is now proposed using the **Java** interface (so, not available with the **GUI** mode) to add as many variables as wished. Of course, these variables will be found in the same list as the initial basic variables.

## New output modes

Previously, using **Java** interface, only three output modes were available:

- `StorageType.MEMORY` ⇒ spacecraft states memory stored
- `StorageType.FILE_SC_BINARY` ⇒ spacecraft states directly stored in a [SQLite format](#) file
- `StorageType.FILE_COLUMNS` ⇒ output variables stored in a [SQLite format](#) file

Now, it is possible to combine any kind of output modes as explained [here](#).

## Some new util methods

New methods have been declared as "public" to read data from files (see [here](#)).

## Anomaly corrections

- Corrections due to [GENOPUS](#) anomalies
  - Impossibility to initialize an orbit using a H0-n frame.
  - Calling [GPVisibilityFromStationDetector](#) or [GPAttitudeSequence](#) widgets make the tool incompatible with a headless mode (no display).
- **PSIMU** is now able to propagate a trajectory using initial orbital parameters correspond to an hyperbolic trajectory.
- when propagation duration is exactly equals to zero, **PSIMU** will no more exit in error.
- use of mean zonal terms was previously wrong in case of *Droziner* or *Cunningham* equations (not available in **GUI** mode); it is now corrected.
- In **Java** mode, using **FILE\_SC\_BINARY** output option, calling [getSpacecraftStateList\(\)](#) or [getSpacecraftState\(i\)](#) methods now gives the same kind of results (sort problem corrected).
- geocentric altitude and latitude (so in **ITRF** frame) are now correctly computed.

# Main differences between V11.2 and V11.3

## Summary

- Upgrade of PATRIUS, GENIUS and GENOPUS versions
- Warning messages when saving result files
- New output variables
- New Ground station management
- Ground track plot
- Anomaly corrections

## Upgrade of PATRIUS, GENIUS and GENOPUS version

This **PSIMU** version is now linked with [PATRIUS V4.1.1](#), [GENIUS V1.8](#) and [GENOPUS V2.0.1](#)

## Warning messages when saving result files

When using [Save Results](#) possibility, if files already exist, a pop up window appears with a warning message asking for overwriting them or not.

## New output variables

This version owns new output variables with:

- solar radiation pressure acceleration (components and norm);
- thrust acceleration (components and norm);
- thrust value;
- mean Lyddane orbital elements
- percentage of the satellite and solar panels illuminated surface.

## New Ground station management

Thanks to the new specific [GENOPUS](#) widget, it is easier to deal with ground stations list.

## Ground track plot

In addition to the [plots](#) already available, a specific panel allows to plot directly the ground track with computed orbital events.

## Anomaly corrections

- Reset of secondary plots: after propagating a new trajectory, all [Output plots](#) plots are reset, even secondary windows if they exist ("+" button on the right side of the panel).
- SQLite file closing: correction of the previous anomaly as the [SQLite](#) file was not correctly closed.
- SQLite bug with LINUX: previously, with some [LINUX](#) configuration (mainly when you have no full rights on /tmp directory), there was a problem when writing [SQLite](#) file. It is now fully corrected.

- Bad line for output files: the **ASCII** output files (ephemeris and/or event files) had a specific line that made these files incompatible with the [GENIUS](#) GPlotpanel widget. This line does not exist anymore.
- Bug correction for AOL events in retropolation: thanks to the new [GENOPUS](#) version (V2.0.1), this bug is now corrected.

# Main differences between V11.1 and V11.2

This version corrects few minor anomalies and is now linked with [PatriusDataset](#) 1.0.3.

## PatriusDataset V1.0.3

This new data set includes:

- new **UTC/TAI** table
- new **JPL** DE406 ephemeris (by default !)
- new **EOP** data

## Anomaly corrections

- Possibility again to build **VTS** files with the **Java** interface (disappeared in V11.1)
- Unit menu for duration of the propagation had disappeared in V11.



# Main differences between V11.0 and V11.1

This version includes essentially ergonomic improvement of the tool. Nevertheless, some new functionalities as [SQLite](#) output files have a real interest.

## Summary

- Compatibility with PATRIUS V3.3
- GUI improvements
- Cartesian coordinates in the Console output
- By default configuration frame
- Null dry mass
- Attitude management
- Improvement of graphical displays
- Variable names
- Output methods
- Events in SpacecraftState list
- Output storage
- Anomaly corrections

## Compatibility with PATRIUS V3.3

This **PSIMU** version is compatible with [PATRIUS](#) V3.3, especially with its new management of exceptions due to divisions by zero as well as corrections when taking into account surface force models in conjunction with continuous thrust (numerical aspects).

## GUI improvements

- Some **Copy&Paste** (and **Import/Export**) functionalities have been added to some widgets as for the orbital initial conditions or the maneuvers.
- For events which manage multiple occurrences, changes occurred to be more obvious for the user with now three entries:
  - From n<sup>th</sup> occurrence
  - Every n occurrence
  - Until n<sup>th</sup> occurrence
- Force models:
  - Choice of the "attraction models" (i.e. **Balmino**, **Cunnigham**, ...) has been hidid with **Balmino** choice by default.
  - By default values are proposed for **SRP**.
- Maneuvers:
  - If engines and/or tanks needed to define a maneuver is set to **NONE**, the maneuver (and the sequence) raised an error.
  - by default, maneuvers components are set to "**Angular**" as event is a "**Relative date**".
  - For the sequence of maneuver, the delays between maneuvers are now hidid.

## Cartesian coordinates in the Console output

Now, this output is only displayed in "Debug mode" (Options menu).

## By default configuration frame

The by default configuration frame has been changed considering only precession-nutation.

## Null dry mass

Now a dry mass set to zero is only raising a warning and no more an error. Moreover, it is possible to extrapolate with such a null value, of course with the condition not to consider force models needing a positive one (as drag or **SRP** forces).

## Attitude management

It is now possible to define an attitude sequence more simply that using switches, using a single attitude law.

## Improvement of graphical displays

Now, it is not necessary to load the current result file. Moreover some improvement has been implemented as a better up to date of the title or the management of discontinuities (for example when plotting latitudes or longitudes).

## Variable names

Some names changed in the **XML** files in order to be more consistent (for example, capitals vs lowercas letters

## Output methods

- In «master mode», the `printFiles()` method allows to create same kind of files as in **GUI** mode.
- The `getSV()` method allows to recover output variables.

## Events in SpacecraftState list

Orbital events are now stored inside the SpacecraftState list computed in master mode.

## Output storage

Rather than to use a binary file as in the previous version, the output of the computation including all computed intermediate parameters (with the output step as defined by the user as well as orbital events) are stored in a **SQLite** formatted file which can be read directly by **PSIMU** plots but also by many other tools (some are free) compatible with this format.

Moreover, when using the master mode, an additional parameter is used to define how output data are stored.

=> incompatibility in the code due to an additional argument in the `propagateInMasterMode()` method. Method `getListOfSpacecraftStates()` is also renamed as `getSpacecraftStateList()`.

## Anomaly corrections

- In the previous version, sometimes the JVM continued to run after exiting the GUI application
- After reading a scenario file, if an error (exception) is raised as the absolute path of the file contains some blank characters, nothing previously occurred (i.e. no messages appeared while data do not have been loaded) => now corrected in V11.1
- bug when displaying perigee altitude.
- bad update of the units for date offsets.